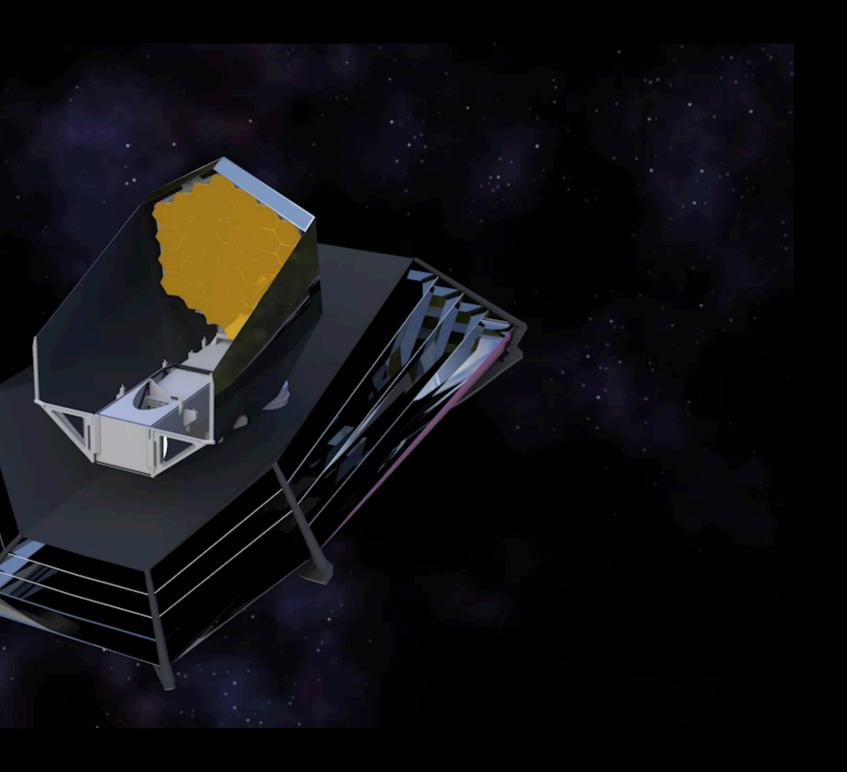


## A R Cooray for the OST STDT

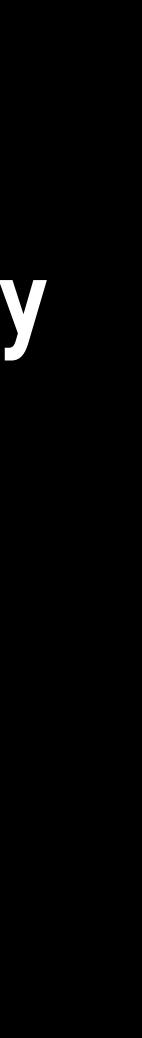
http://origins.ipac.caltech.edu

# **ORIGINS** Space Telescope





## @NASAOriginsTele

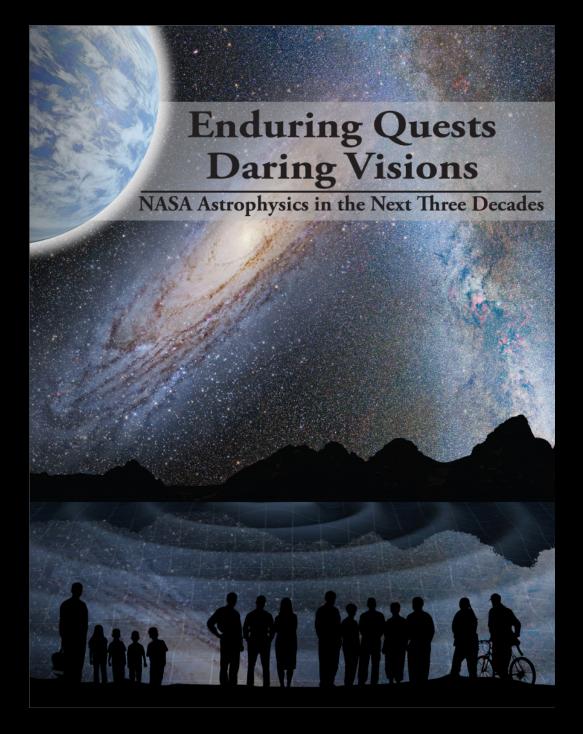




NASA flagship class mission concept for the 2020 Decadal review. **Comes from the NASA Astrophysics Roadmap.** 

- <6  $\mu$ m 600  $\mu$ m (diffraction limit around 20-40  $\mu$ m)
- 4.5-5K actively-cooled 8-13m aperture operating at L2
- large gain in sensitivity => new spectroscopic capabilities
- exoplanet study capabilities via a mid-IR coronagraph
- modular instrument suite with robotic serviceability at L1
- Mission aimed at mid 2030s: post JWST, concurrent with WFIRST, Athena, LISA, and 25m-35m ground-based optical/IR facilities.
- Science goals and measurement requirements in 2030+





# **ORIGINS** From the first stars to life Space Telescope



- NASA Appointed Members: L. Armus, IPAC; C. Battersby, UConn; J. Bauer, UMD; E. Bergin, Michigan; M. Bradford JPL; K. Ennico-Smith, Ames; UMass; T. Roellig, Ames; K. Sandstrom, UCSD; K. Stevenson, STScl; K. Y. L. Su, Arizona; J. Vieira, UIUC; E. Wright, UCLA; J. Zmuidzinas, Caltech
- Ex-officio representatives: S. Neff & E. Smith, NASA Cosmic Origins Program Office; S. Alato, SNSB; D. Burgarella, LAM, France; D. Scott, CAS; Gerin, CNES; I. Sakon, JAXA; F. Helmich, SRON; R. Vavrek, ESA; K. Menten, DLR; YS Song, KASI; S. Carey, IPAC
- Technologist), L. Fantano (Thermal Systems Engr), A. Jones (Mechanical Systems Engr), J. Howard (Optical Systems Engr), J. Corsetti (Optical Engr), E. Canavan (Cryo Engr), J. Staguhn (Instrument Scientist)
- Lipscy, Ball; J. Mather, GSFC; H. Moseley, GSFC; G. Rieke, Arizona; M. Rieke, Arizona; J. Turner, UCLA; M. Urry, Yale.



## **Study Team**

 Community Chairs: A. R. Cooray, UCI; M. Meixner, STSCI/JHU • Study Scientist: D. Leisawitz, GSFC • Deputy Study Scientist: J. Staguhn, GSFC/JHU • Study Manager: R. Carter, GSFC • NASA HQ Program Scientists: K. Sheth, D. Benford

J. Fortney, UCSC; L. Kaltenegger, Cornell; G. Melnick, CfA; S. Milam, GSFC; D. Narayanan, UFlorida; D. Padgett, JPL; K. Pontopiddan, STSCI; A. Pope, • NASA Study Center (Goddard Space Flight Center) Team: A. Flores (Mission Systems Engr), J. Kellogg (Instrument Systems Engr), M. DiPirro (Chief)

• Study Advisory Board: J. Arenberg, Northrup Grumman; J. Carlstrom, Chicago, H. Ferguson, STScl, T. Greene, Ames; G. Helou, IPAC; C. Lawrence, JPL; S.

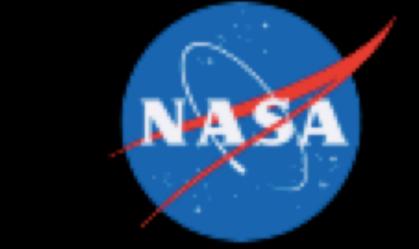
Μ.





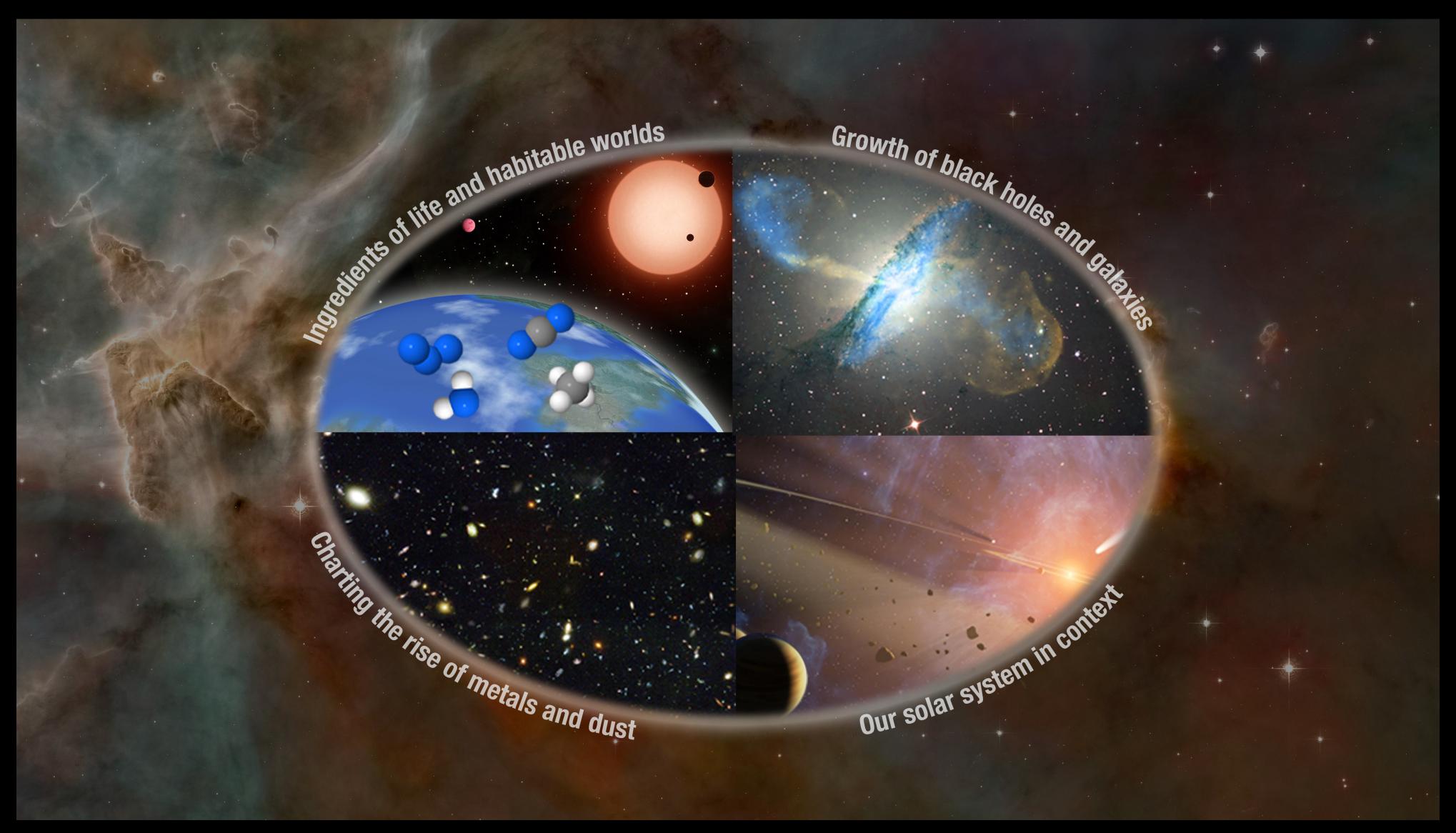


- Solar System: Stefanie Milam, James Bauer
- Planet Formation: Klaus Pontoppidan and Kate Su
- Exoplanets: Kevin Stevenson, Jonathan Fortney
- Milky Way and Nearby Galaxies: Karin Sandstrom and Cara Battersby
- Galaxy Evolution over Cosmic Time: Lee Armus and Alex Pope
- Early Universe and Cosmology: Matt Bradford and Joaquin Vieira



**Science Working Groups** 









## Searching for biosignatures in nearby exoplanets

Ozone **(O**<sub>3</sub>)

10µm

Methane  $(CH_4)$ 

5µm

Water **Carbon Dioxide**  $(CO_2)$ 

20µm



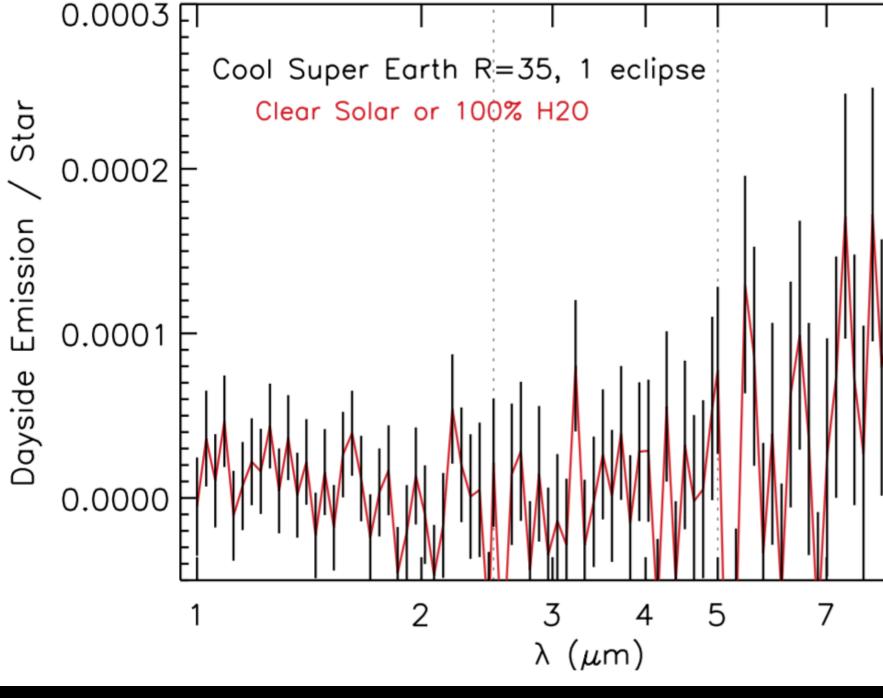


## To detect biosignatures:

- Spectral resolving power ( $\lambda/\Delta\lambda$ ) of 30-50
- Noise floors < 10 ppm 0
  - $-(M3V@20 pc 2 hr at 7 \mu m)$
- Key spectral signatures of Super-Earths that Origins will detect:
  - $-9 \mu m$  for ozone (biosignature)
  - $-7 \mu m$  for methane (life detection)

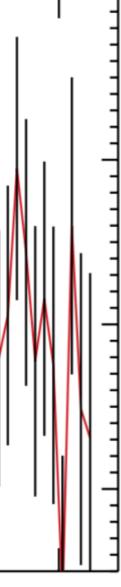
Origins Space Telescope will have mid-IR capability down to 6 µm; noise floor will be due to mid-IR detector stability.

### At 30ppm-50ppm JWST cannot study habitable zone worlds (Greene et al. 2016)





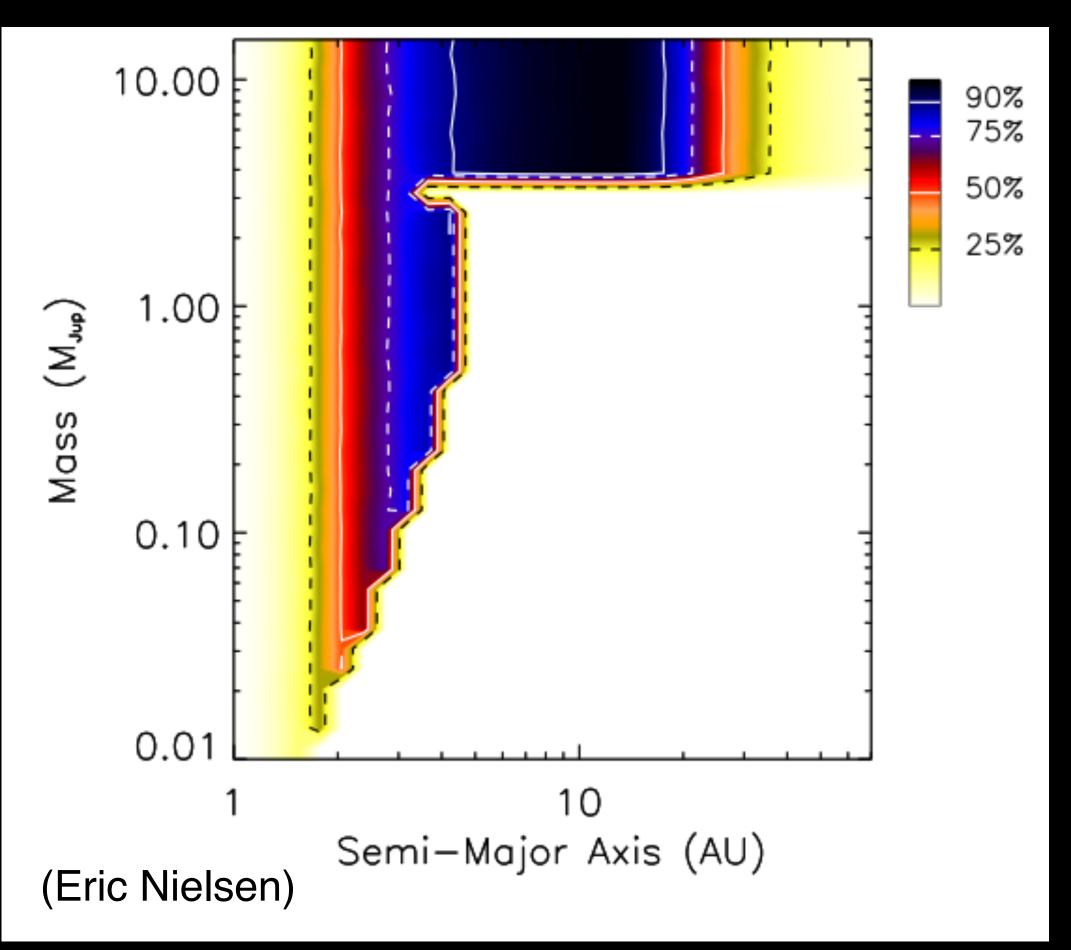






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Directly image warm Neptunes and Jupiters around the nearest Sun-like Stars

Coronagraph will enable direct imaging of Jupiters at 5 – 14 AU and warm Neptunes into 2 AU

Kepler finds planets smaller than Neptune are ubiquitous close to their parent stars.

Near the habitable zone of the closest stars, the thermal emission of these planets can be bright enough to be seen behind the glare of their parent stars.

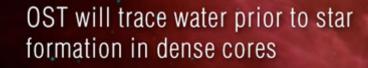
OST spectroscopy will allow us to directly probe the atmosphere and composition of these "Neptunes".







## Following the trail of life-bearing water from the interstellar medium to habitable worlds\*



OST will follow the trail of water into nascent planet-forming disks





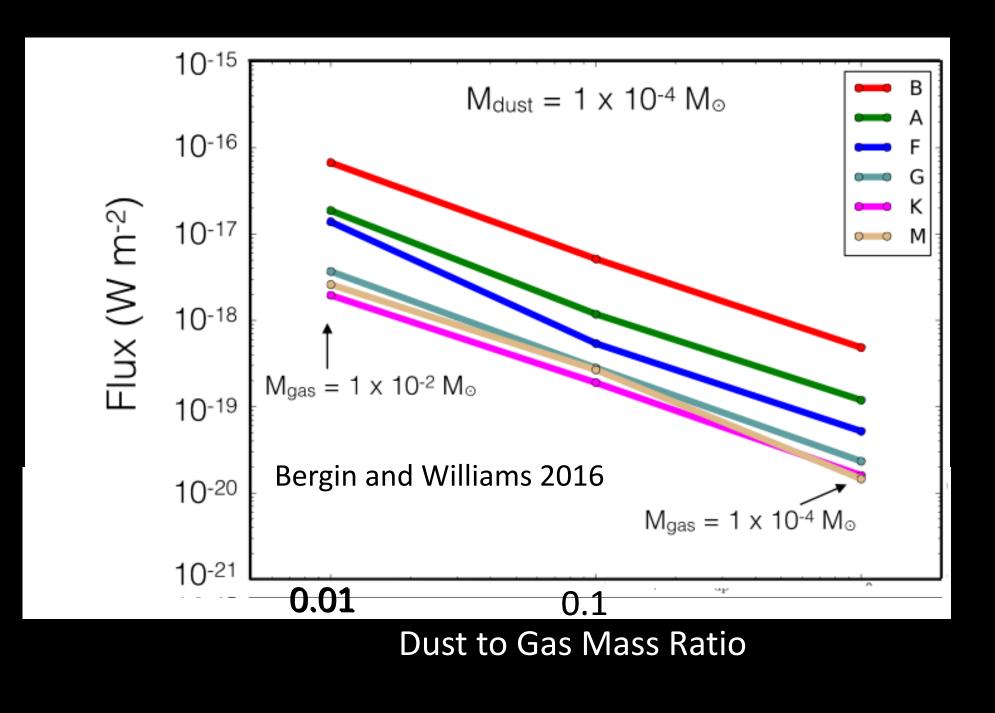
OST will survey thousands of disks and reveal the statistical disposition of water around stars of all masses during planet formation

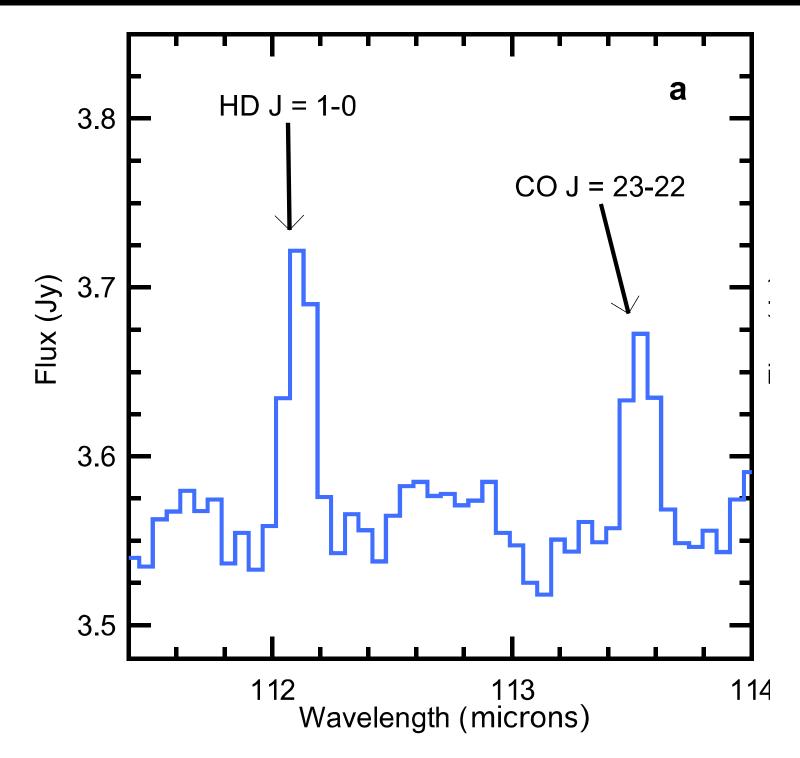
OST will set distinct constraints on planetary habitability by detecting water and biomarkers on rocky planets in habitable zones

# Space Telescope From the first stars to life

## Probing the total gas content during the time of planet formation

What are the timescales of gas/ice giant and super-Earth formation? What is the total gas content to unlock the ability to follow the implantation of C, H, O, N into pre-planetary materials?. Use HD to measure the gas mass in disks down to cool stars with a gas/dust mass ratio of unity.





Herschel Detection of HD J = 1-0 towards TW Hya providing the first (semi)direct contraints on the gas mass (Bergin et al. 2013)



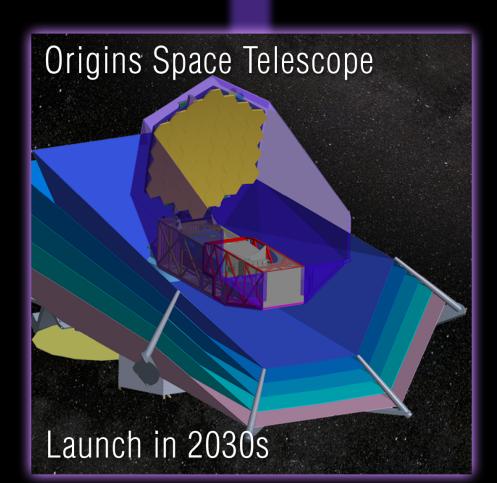


# Seeing into the dark ages with Origins Space Telescope (OST)

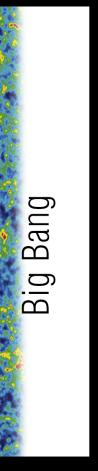
Today			Reioni	zation First Galaxies	First Stars	Dark Ages		
REDSHIFT	1	6	8	12			16	1100
BILLIONS of	YEARS AGO	12.6	13.1	13.4			13.5	13.8



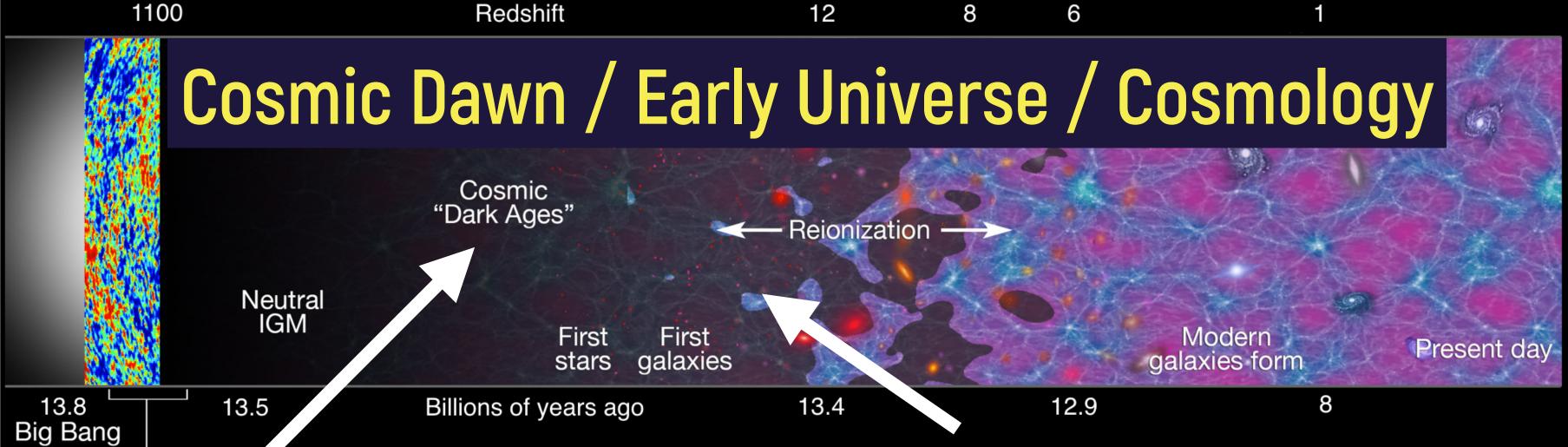
James Webb Space Telescope Launch in 2019







# **ORIGINS** From the first stars to life Space Telescope



Origins goes further!

- collapsing to form first stars! Primordial cooling via H2 rotational lines
  - Seeds of super massive black holes

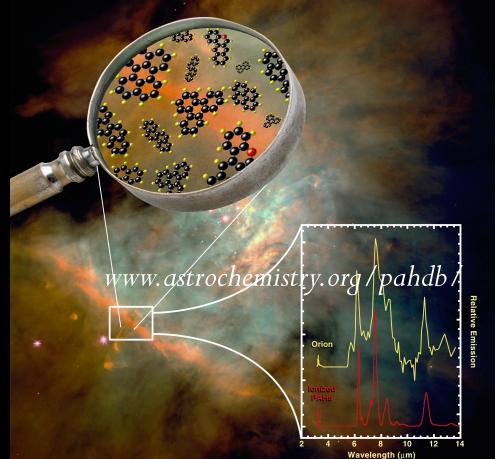
To detect primordial H<sub>2</sub> line cooling at formation sites of first stars and galaxies at z ~10-15 Origins Space Telescope sensitivity will need to be down to 10-23 Wm<sup>-2</sup> in a deep field integration in rotational lines (rest-frame 12.3,17, 28 µm)

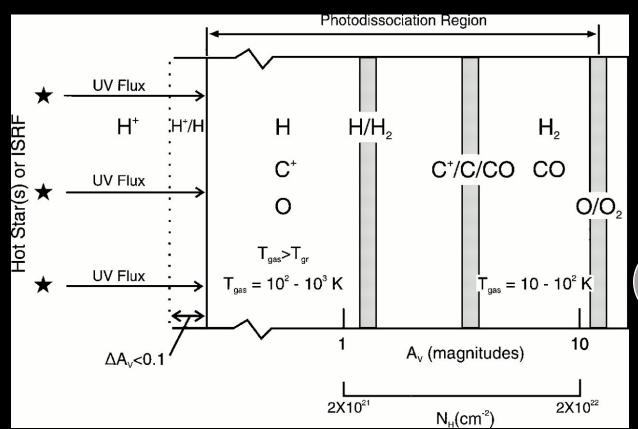


JWST/WFIRST capability is detecting first stellar emission

Origins Space Telescope will venture beyond JWST and image gas

### **ORIGINS** Space Telescope From the first stars to life How do we probe the interstellar medium in high redshift galaxies? The Largest Interstellar Molecules... 20 www.astrochemistry.org/pahdb ) Offset (' Declination 6 PAH Photodissociation Region $H_2$ -60 UV Flux H/H H, C<sup>+</sup>/C/CO CO UV Flux -80 -100 $\mathsf{T}_{\mathsf{gas}}\mathsf{>}\mathsf{T}_{\mathsf{gr}}$ -40 -60 Molecular cloud UV Flux nsion Offset (") $T_{gas} = 10^2 - 10^3 K$ $T_{max} = 10 - 10^2 \, \text{K}$ $\leftarrow$ 10 A<sub>v</sub> (magnitudes) ΔA<sub>v</sub><0.1 Hollenbach & Tielens 1997



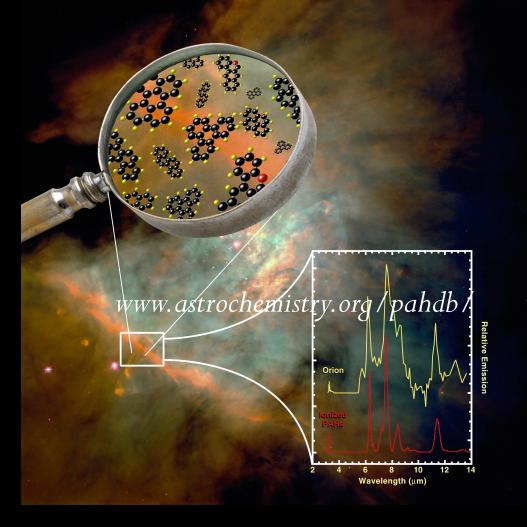




# **ORIGINS** From the first stars to life Space Telescope

## How do we probe the interstellar medium in high redshift galaxies?

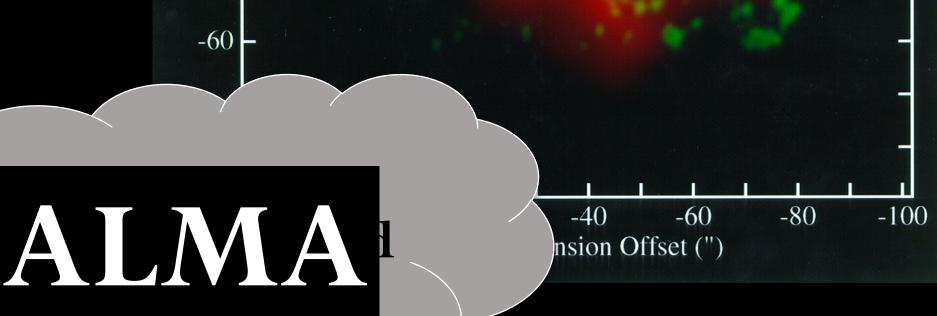
The Largest Interstellar Molecules...





20

# Origins Space Telescope





Hollenbach & Tielens 1997

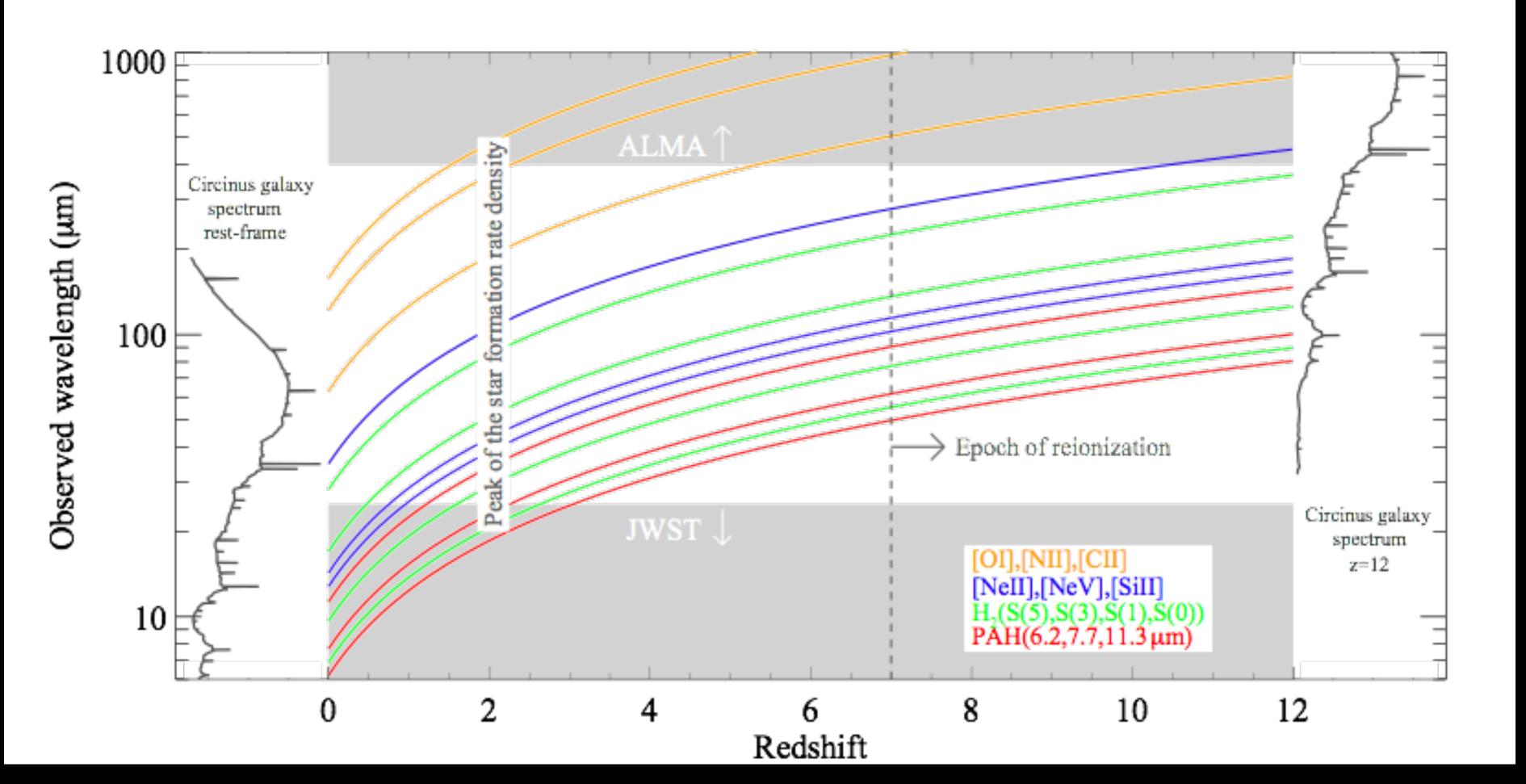
 $H_{2}$ 

WST





## Infrared is rich in key spectral lines!





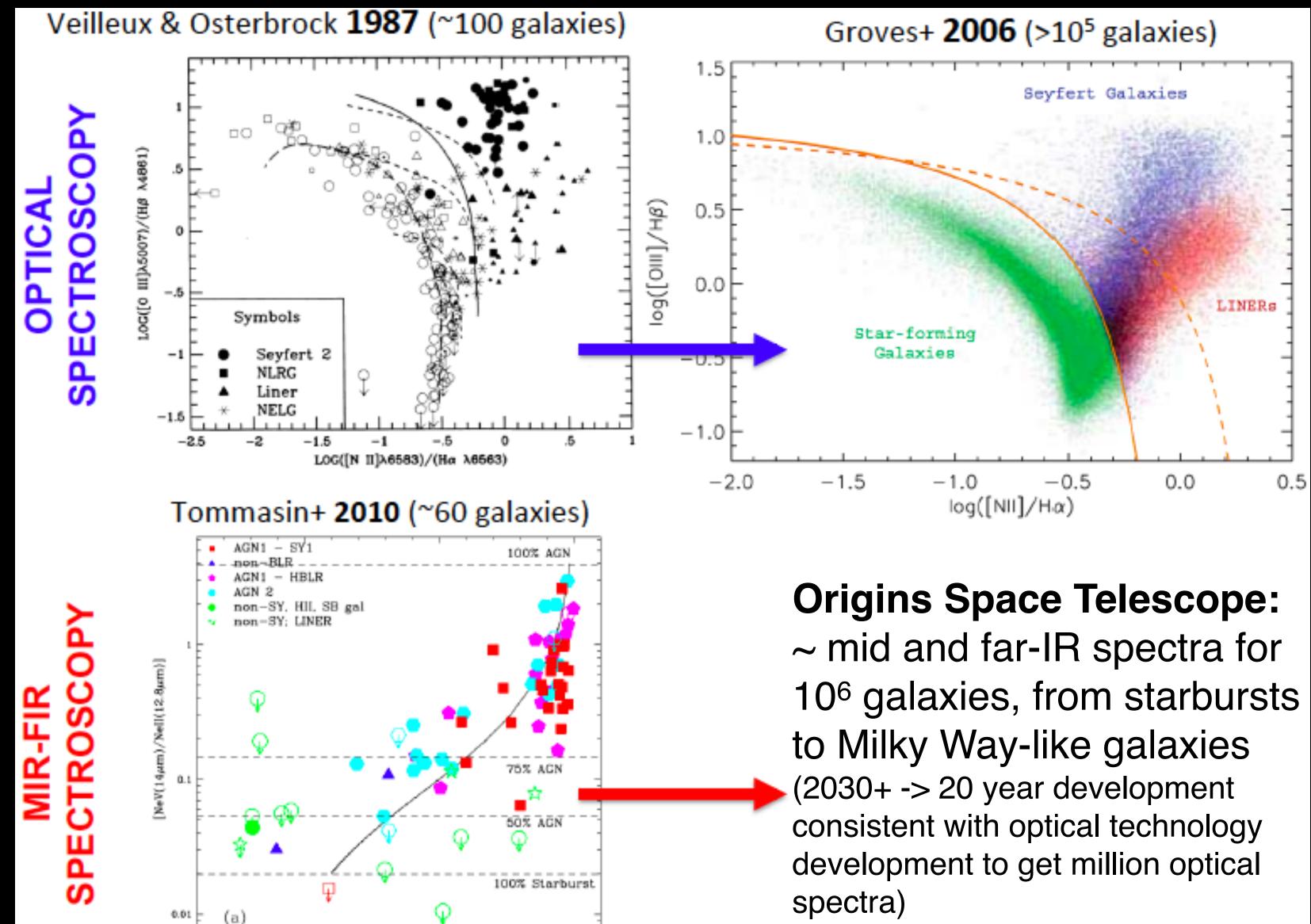
















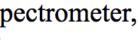


# **ORIGINS** From the first stars to life Space Telescope

### An Example Science Traceability Matrix

OST Science Case	OST Science Theme NASA Science Goal Decadal Science Goal	Salanaa Ohiaatiwaa	Science Requirements		Instrument Requirements			
Number/Title Theme		Science Objectives	Science Observable	Measurement Requirement	Technical Parameter	Technical Requirement	Instrum	
Trace the dust and metal enrichment history of the early Universe. Find the first cosmic sources of dust, and search for	<b>OST-2:</b> (Charting the) Rise of Metals, Dust, and the First Galaxies <b>NASA-2:</b> How did we get here?	z=3 to 0.1 dex down to 10 <sup>11</sup> L <sub>sun</sub> ; (b) determine the cosmic metal abundance Ωmetals from z=0 to z=8 to 0.1 dex accuracy in 8 redshift bins; and (c) measure the multiple phases of the ISM to infer the physical phenomena that regulate SF efficiency at the peak of cosmic star formation at z=1-3.	tracer: [NeII]12.8, [NeIII]15.6, [SIII]18.7, [SIV]10.5; z=0-8 relative metallicity tracer: [OIII] 52+88 $\mu$ m, [NIII] 57 $\mu$ m ; cooling and heating of the ISM through [OI], [OIII], [NII], [CII].	Rest-frame mid and far-IR spectral mapping to select z=0 to 8 galaxies Identify galaxies in a tiered	Wavelength range Spatial resolution	20-800 μm 5 arcsec at 200 μm		
				<ul> <li>57 μm ;</li> <li>cooling and heating of the ISM through [OI], [OIII], [NII], [CII].</li> <li>A multi-tiered survey, with a wide tier of ~10 deg<sup>2</sup>, with sensitivity</li> </ul>	spectral mapping survey Measure line flux densities of identified galaxies	Spectral line sensitivity	(min. 9 m Telescope) 1e-21W m-2 (driven by the MIR lines)	incoherent spec low res mode
					Spectral Resolving power	$\lambda/\Delta\lambda = 500$		
					survey area, instan- taneous FOV, FoR	10 deg^2		
					Mapping Speed			







Identify visionary, robust, and compelling science questions

Derive from those questions a set of high-priority measurement requirements for the mission

Choose a mission architecture

Determine technology needs

**Evaluate trades and iterate engineering design with STDT** 

Estimate cost

**Present to Decadal Survey** 







# (a) ORIGINS Space Telescope From the first stars to life

- Telescope type: three mirror anastigmat; unobstructed primary mirror
- Primary mirror: 9.1 meters in diameter; 37 hexagonal segments
- Five instruments housed in an Instrument Accommodation Module (IAM)

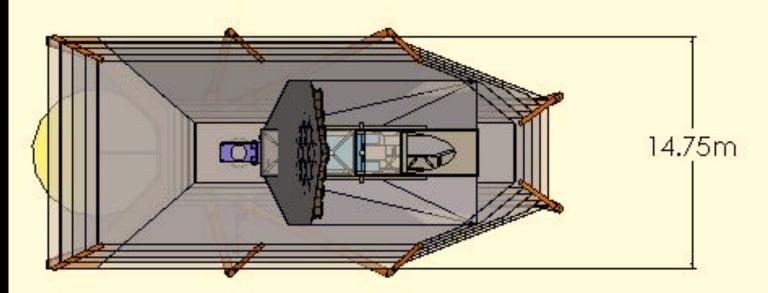
   Medium Resolution Survey Spectrometer (MRSS) JPL
   Hi Res (Far-IR) Spectrometer (HRS) GSFC
   Heterodyne Instrument (HERO) CNES
   FIR Imager/ Polarimeter (FIP) GSFC
   MID-IR Imager Spectrometer/ Coronagraph (MISC) JAXA
- Instrument Wavelength Coverage: 5 to 660 μm
- MISC serves as guider for the spacecraft attitude control system
- Telescope and instrument operating temperature: 4 to 4.5 K
- Cryocoolers (current SOA) used for cooling, not expendable cryogen
- Instrument warm electronics housed in the spacecraft bus (270 K)

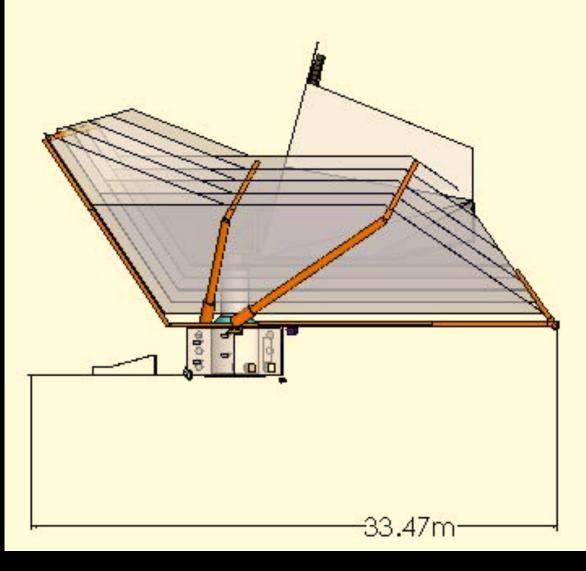


## Concept 1 Highlights



### Deployed

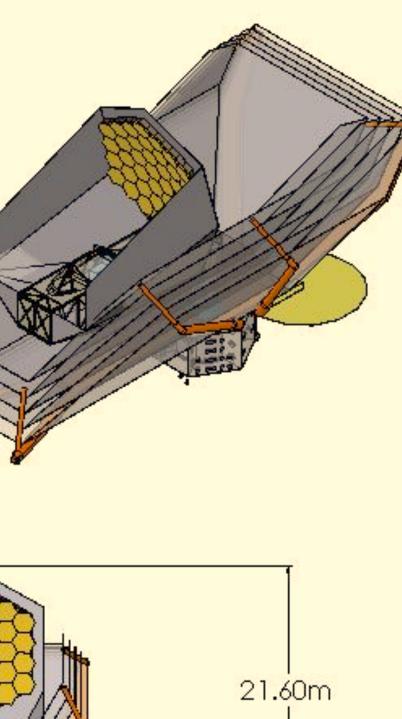




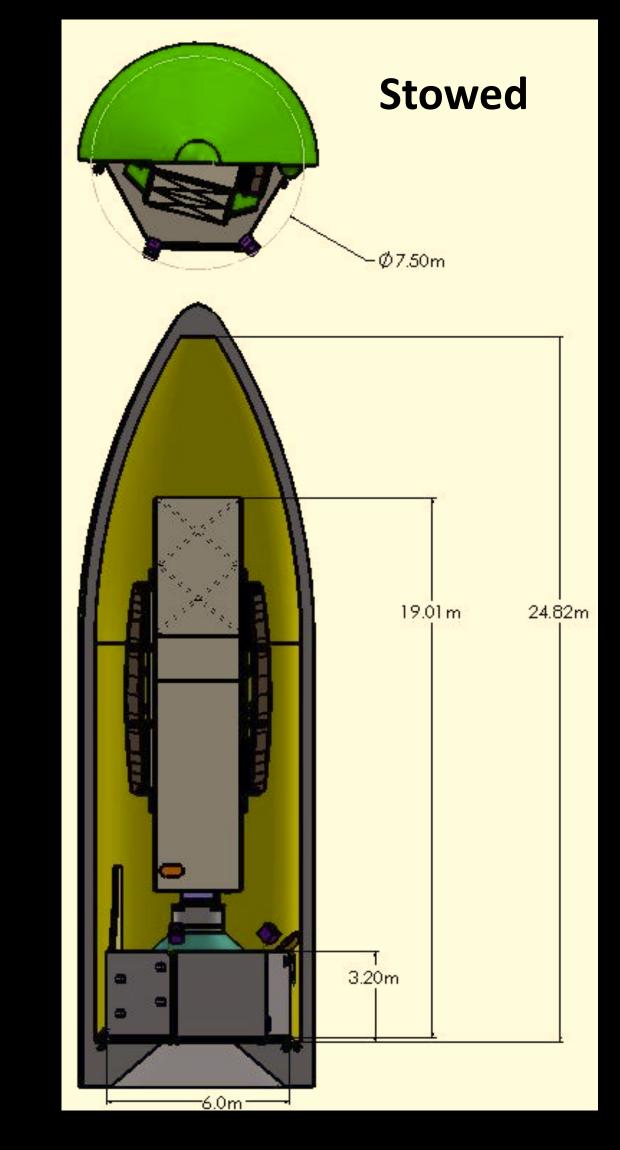


## From the first stars to life Concept 1 configuration





3.5m





- **Mission Life**: 5 Years with 10-year consumables
- Launch Vehicle: SLS Block 2, 8.4m x 27.4m fairing
- LRR: September 1, 2035
- OST Observatory Size:
  - 14.75 x 21.6 x 33.5 m (deployed), 19L x 7.5D m (stowed)
- **Mission Orbit**: Sun-Earth L2 (Sun, Earth, Moon avoidance, No eclipses)
- Pointing Control 44 mas; Pointing Knowledge 30 mas; Jitter 22 mas
- Folded/scooped sunshade to minimize size (size fixed for this study)
- IAM is to be on-orbit serviceable (underside)
- Science Observation: > 70%
- off max power roll angle
- **Communication**: 2 optical terminals, 1 S-band OMNI Pair, 1 S-band HGA
- **Observatory Mass**: ~30000 kg (CBE)
- Observatory Power: ~7500 W (CBE)
- Peak Data Rate: ~350 Mbit/sec

From the first stars to life **Concept 1 Requirements** 



• Field-of-Regard (FOR): -5°- +45° Pitch off Sun Line, 360° Yaw about Sun Line, ±5° Roll about Line of Sight (LOS)

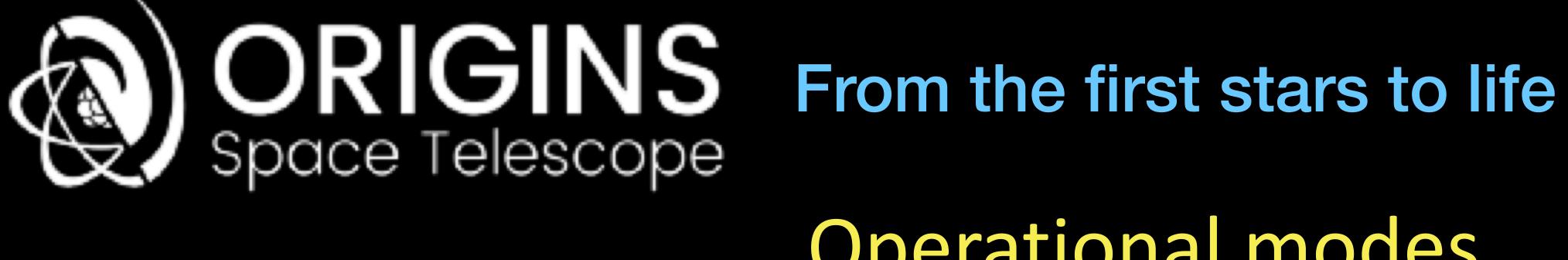


**Instrument Specifications** 

Instrument	Wavelength	Resolving Power	 Typical Required Sensitivity:	Other
MISC		imager: R~15; spectrometers, R=300 to 25000	photometric: 1 μJy @10 μm	coronagraph 10 <sup>-7</sup> -10 <sup>-8</sup> IWA=λ/D
	40, 80, 120, 240 μm	R~15	1 μJy - 10 mJy (confusion limit)	polarimetry, spectral line filters
MRSS		low-res~500 high-res~4x10 <sup>4</sup>	2.6x10 <sup>-21</sup> W/m <sup>2</sup> (spectral line)	4-5 channels
HERO	150 to 600 µm	~107	2 mK in 0.2 km/s @ 1 THz	polarized, background limited
HRS		low-res ~ 5x10 <sup>4</sup> high-res~5x10 <sup>5</sup>	2.6x10 <sup>-21</sup> W/m <sup>2</sup> 5 $\sigma$ (spectral line)	photo-counting

s to life		NASA
cal Required itivity:	Other	





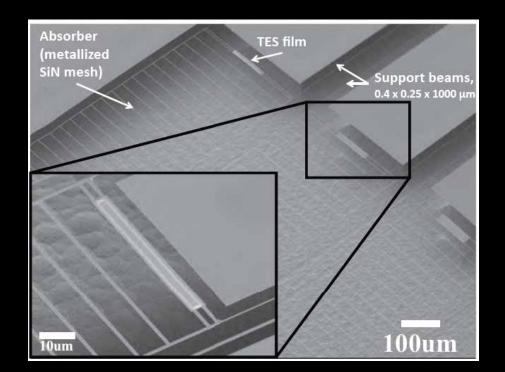
- Large scale survey mapping: FIP, MRSS, MISC (100 arcsec/second scan speed allowing 1000 sq. degree+ survey maps with FIP etc)
- Small maps: HERO, MISC, FIP, MRSS, HRS
- Pointed Observations: HERO, MISC, FIP, MRSS, HRS
- Coronagraphy: MISC
- Transit spectrometer: MISC

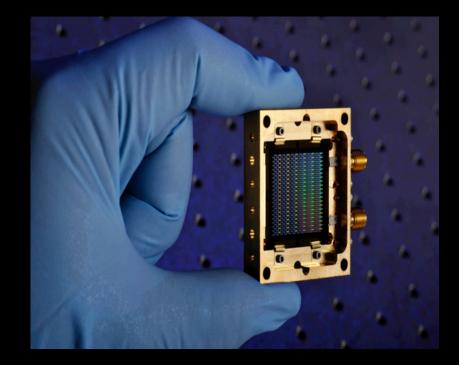


## **Operational modes**

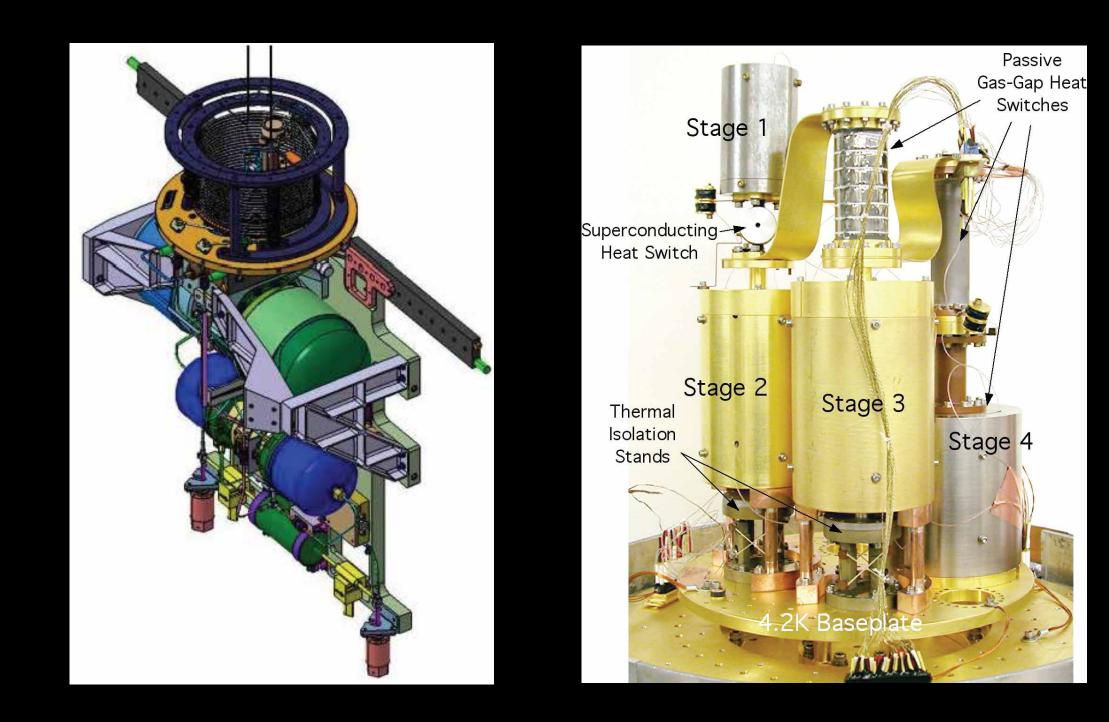
# **ORIGINS** From the first stars to life Space Telescope Technology Gaps

- Compact Far-IR spectrometers
- Heterodyne detectors
- Sub-Kelvin cooling
- Large cryogenic optics and actuators
- 4.5 K cryocoolers
- Mid-IR detectors and coronagraphy





### • Large-format, high-sensitivity far-IR direct detectors, multiplexers, and readout electronics





# **ORIGINS** From the first stars to life Space Telescope

## What's coming up in 2018+

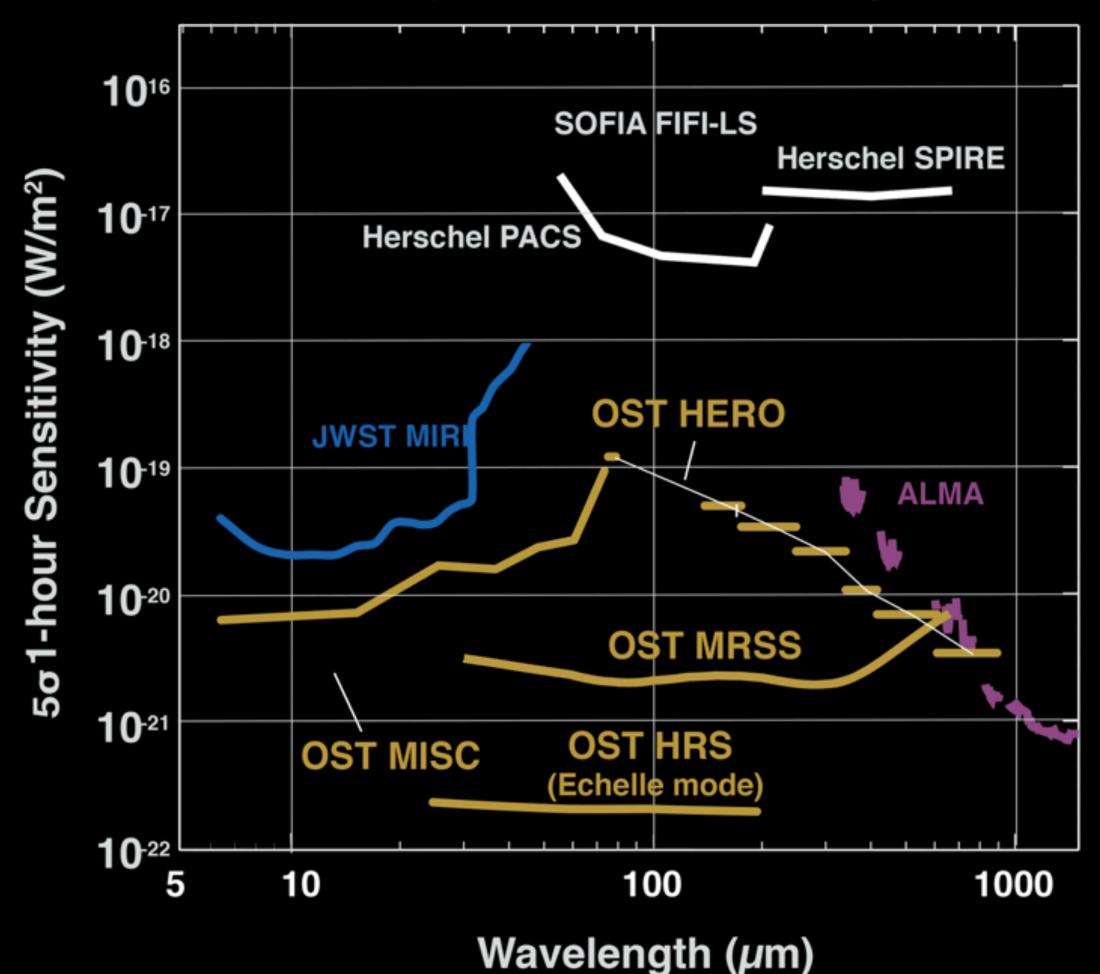
- Concept I complete STDT will deliver a report to NASA this Spring Concept 2 definition exercise started.
- Define Concept 2 and assess engineering feasibility
  - •Aperture size
  - Instrument suite and priority
  - Inform/communicate feasibility assessments to the STDT
  - Iteration with the STDT
- Select Concept 2 criteria: STDT has agreed to consider a JWST-like size design that will fit into a maximum of a 7m-class fairing

Concept 2 will be less ambitious than Concept 1, esp instrument capabilities. Likely will be limited to 3 de-scoped instruments developed for Concept I.





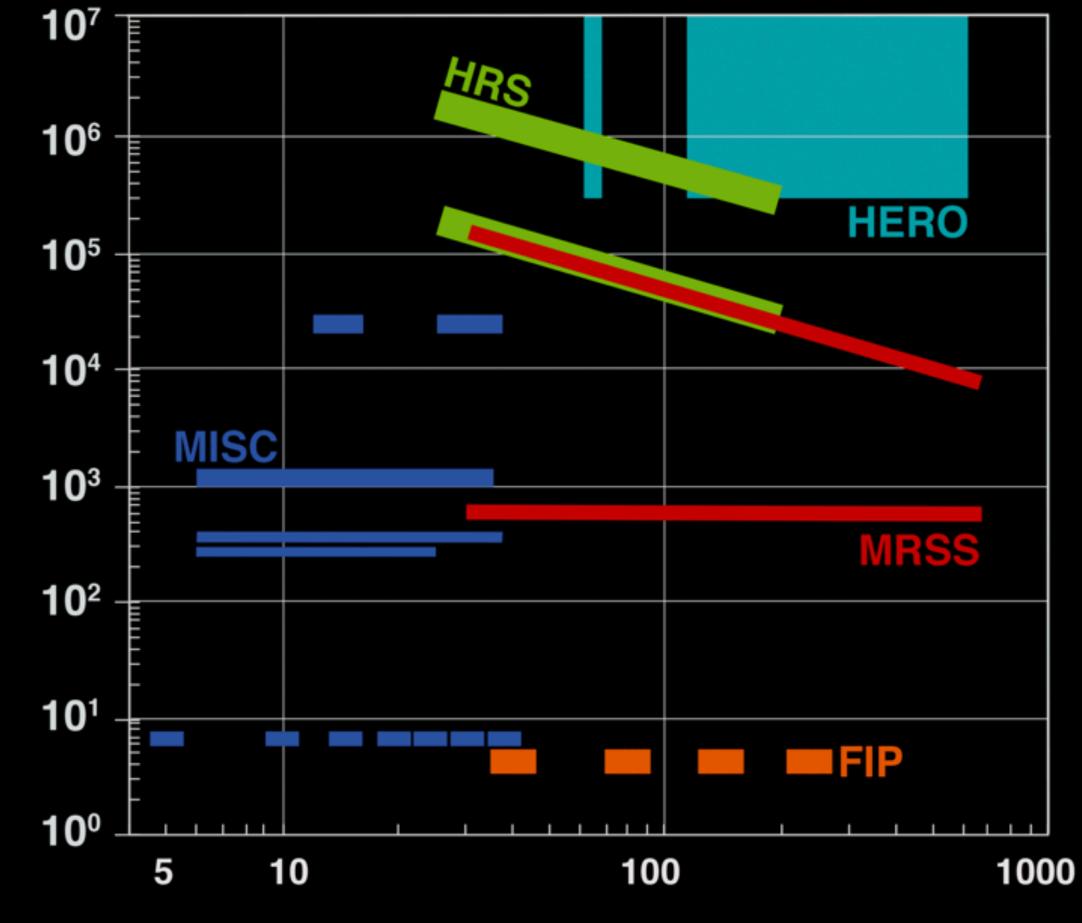
### Spectral Line Sensitivity



Spectral Resolving Power (MAA)



### **Spectral Resolution**



Wavelength (µm)



## What Origins Space Telescope will do

- Study gas cloud cooling at cosmic dark ages, to ozone and methane our Solar system.
- Provides a factor of 10,000 (!) improvement in sensitivity. An immense discovery potential.
- true revolution in astronomy.



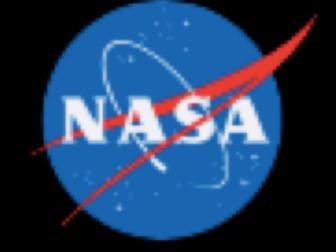
biosignatures of exoplanets, to pathway of water to habitable exoplanets and

Origins Space Telescope will not be extending what we know already. It will be a



## What Origins Space Telescope will be

- We want to hear about your:
  - Scientific questions that would define and use such an observatory
  - Your technical innovations that would help make *Origins* a reality. 0



### A flagship general observatory - community driven sciences and instruments.